

# RADIOCOMMUNICATION STUDY GROUPS

Document 4/94(Rev.1)-E 6 October 2006 English only

Source: Document 4A/TEMP/204

Subject: Question ITU-R 236/4

# Working Party 4A

#### DRAFT REVISION OF RECOMMENDATION ITU-R S.1586

# Calculation of unwanted emission levels produced by a non-geostationary fixed-satellite service system at radio astronomy sites

(Question ITU-R 236/4)

(2002)

#### **Summary**

In this draft revision of Recommendation ITU-R S.1586 the mathematical model of a radio telescope antenna gain pattern in Annex 1 is replaced by a reference to the antenna pattern described in Recommendation ITU-R RA.1631. Also, in Annex 1, the method for determining the epfd distribution in the worst-case pointing direction of a radio astronomy antenna in respect of interference from a non-GSO satellite system is modified to take into consideration the 2%-of-noise criterion in Recommendation ITU-R RA.1513-1.

#### DRAFT REVISION OF RECOMMENDATION ITU-R S.1586

# Calculation of unwanted emission levels produced by a non-geostationary fixed-satellite service system at radio astronomy sites

(Question ITU-R 236/4)

(2002)

### **Scope**

This Recommendation describes a method that could be used to calculate the unwanted emission levels produced by a non-GSO fixed-satellite service system on radio astronomy sites. It also contains a procedure for the calculation of the percentage of time during which a given equivalent power flux-density (epfd) is exceeded when the receiving antenna gain is assumed to be 0 dBi in the direction of the incoming interference, and a given integration time is considered.

The ITU Radiocommunication Assembly,

considering

- a) that, in some cases, the radio astronomy service and space services (space-to-Earth) have been allocated to adjacent or nearby frequency bands;
- b) that the radio astronomy service is based on the reception of emissions at much lower power levels than are generally used in other radio services;
- c) that, due to these low received power levels, the radio astronomy service is generally more susceptible to interference from unwanted emissions than other services;
- d)\* that Recommendation 66 (Rev.WRC-2000) requestsed in recommends 5 that ITU-R 5 the ITU-R to "study those frequency bands and instances where, for technical or operational reasons, more stringent spurious emission limits than the general limits in Appendix 3 may be required to protect safety services and passive services such as radio astronomy, and the impact on all concerned services of implementing or not implementing such limits;";
- e)\* that Recommendation 66 (Rev.WRC-2000) requestsed in recommends 6 that ITU-R 6 the ITU-R to "study those frequency bands and instances where, for technical or operational reasons, out-of-band limits may be required to protect safety services and passive services such as radio astronomy, and the impact on all concerned services of implementing or not implementing such limits;";
- f)\* that several footnotes to the Radio Regulations (RR) (such as RR Nos. 5.149, 5.443B; and 5.511A-and 5.551G) draw attention to the protection of the radio astronomy service, particularly from space-borne transmitters;
- g) that due to the characteristics of non-geostationary (non-GSO) satellite systems, and in particular to the time-varying nature of interference, the level of interference from such satellites into radio telescopes cannot be evaluated in the same way as for GSO satellites,

<sup>\*</sup> Recommendation 66 (Rev.WRC-2000) and RR No. 5.551G were suppressed by WRC-03.

#### recommends

- that the calculation of unwanted emission levels produced by a non-GSO fixed-satellite service (FSS) system on radio astronomy sites could be conducted by administrations using the method described in Annex 1;
- that when performing these calculations, the antenna pattern described in <u>Recommendation</u> ITU-R RA.1631<del>Annex 2</del> could be used to model radio astronomy antennas;
- that the percentage of time during which an equivalent power flux-density (epfd) level (defined assuming a 0 dBi receiving antenna gain in the direction of interference and given an integration time) is exceeded could be calculated according to the method described in Annex 23.

#### Annex 1

# Calculation of unwanted emission levels produced by a non-GSO FSS system at radio astronomy sites

The methodology described here, based on the epfd concept defined in RR Article 22, No. 22.5C, is intended for use in calculating the power flux-density (pfd) levels produced by unwanted emissions of a non-GSO FSS satellite system into radio telescopes, taking into account the characteristics of both the satellite system and the radio telescope antenna. The value of the epfd is the aggregate of the contributions from all satellite emissions expressed as the pfd of a single equivalent source on the boresight (peak of main beam) of the radio telescope.

#### 1 Required parameters

Due to the particular characteristics of non-GSO satellite systems, it is clear that the level of the interference from such satellites into a radio telescope cannot be evaluated in the same way as for GSO satellites. A statistical approach is needed which takes into account the dynamic aspect of non-GSO satellites.

The evaluation of interference resulting from the satellites at the radio telescope during the integration time (2000 s) should be based on statistical calculations and should take into account the parameters of both the satellites and the radio telescope.

Non-GSO satellite system parameters:

- the number of satellites visible in the sky at the radio astronomy station;
- the pfd at the radio telescope within the radio astronomy band considered, estimated using a dBsd or dBc mask;
- the distances between the satellites and the radio astronomy station;
- the detailed orbital characteristics of the satellites.

#### Radio telescope parameters:

- the antenna location;
- the antenna pattern and antenna gain;
- the practical range of pointing directions;

- the boresight pointing direction;
- the off-axis angles between the boresight of the antenna of the radio astronomy station and the directions of the transmitting satellites;
- the integration time (2000 s).

### 2 Calculation of epfd levels at radio astronomy sites

The receiving gain of a radio telescope in the direction of a non-GSO satellite (as opposed to a GSO satellite) varies with time chiefly because of the movement of the satellite and the fine angular structure of the radio telescope's side-lobe pattern. There will be times when the telescope gain in the direction of a satellite is much higher than 0 dBi, and other times when it is less. In addition, in the case of multiple satellites of a non-GSO system, all their contributions must be included and properly taken into account.

This may be done using the concept of epfd originally defined to assess possible sharing conditions between GSO and non-GSO systems. In the section below the concept is developed for the case of a radio astronomy station subject to interference from non-GSO satellites. The definition is based upon RR No. 22.5C as adopted at the World Radiocommunication Conference (Istanbul, 2000) (WRC-2000).

#### 2.1 Definition of epfd

When an antenna receives power, within its reference bandwidth, simultaneously from transmitters at various distances, in various directions and at various levels of incident pfd, the epfd is that pfd which, if received from a single transmitter in the far field of the antenna in the direction of maximum gain, would produce the same power at the input of the receiver as is actually received from the aggregate of the various transmitters.

The instantaneous epfd, expressed in  $dB(W/m^2)$ , is calculated using the following formula:

$$epfd = 10 \log_{10} \left[ \sum_{i=1}^{N_a} 10^{\frac{P_i}{10}} \cdot \frac{G_t(\theta_i) - G_r(\varphi_i)}{4\pi d_i^2} \right]$$
(1)

$$epfd = 10\log_{10} \left( \sum_{i=1}^{N_a} 10^{\frac{P_i}{10}} \cdot \frac{G_t(\theta_i)}{4\pi d_i^2} \cdot \frac{G_r(\varphi_i)}{G_{r,max}} \right)$$
(1)

where:

 $N_a$ : number of non-GSO space stations that are visible from the radio telescope

*i*: index of the non-GSO space station considered

- $P_i$ : RF power of the unwanted emission at the input of the antenna (or RF radiated power in the case of an active antenna) of the transmitting space station considered in the non-GSO system in the reference bandwidth (dBW)
- $\theta_i$ : off-axis angle between the boresight of the transmitting space station considered in the non-GSO system and the direction of the radio telescope  $G_t(\theta_i)$ : transmit antenna gain (as a ratio) of the space station considered in the non-GSO system in the direction of the radio telescope
- $d_i$ : distance (m) between the transmitting station considered in the non-GSO system and the radio telescope

 $\varphi_i$ : off-axis angle between the pointing direction of the radio telescope and the direction of the transmitting space station considered in the non-GSO system

 $G_r(\varphi_i)$ : receive antenna gain (as a ratio) of the radio telescope, in the direction of the transmitting space station considered in the non-GSO system (see Recommendation ITU-R RA.1631Annex 2)

 $G_{r,max}$ : maximum gain (as a ratio) of the radio telescope

*epfd*: instantaneous epfd in the reference bandwidth at the radio telescope  $(dB(W/m^2))$ .

The epfd calculation in equation (1) assumes that the pfd due to all interfering sources is directed at the boresight of the receiving antenna, where the antenna gain is maximum. However, radio astronomy protection criteria are based on a 0 dBi contour of the radio astronomy antenna. Using the approach in equation (1), tThe pfd due to all interfering sources directed at the 0 dBi gain of the receiving antenna, can be determined as follows:

• From equation (1), the instantaneous epfd directed at the 0 dBi gain of the receiving antenna, expressed in (W/m²), is given by

$$\frac{epfd_{G_r = 0 \text{ dBi}} = 10 \log_{10} \left[ \sum_{i=1}^{N_a} 10^{\frac{P_i}{10}} \cdot \frac{G_t(\theta_i)}{4\pi d_i^2} \cdot \frac{G_r(\phi_i)}{G_r(\phi_i)} \right]}{(2)}$$

$$epfd_{G_r = 0 \, dBi} = \left( \sum_{i=1}^{N_a} 10^{\frac{P_i}{10}} \cdot \frac{G_t(\theta_i)}{4\pi \, d_i^2} \cdot G_r(\varphi_i) \right)$$
 (2)

The instantaneous  $epfd_{G_r=0 \text{ dBi}}$  values resulting from equation (2), averaged over a 2000 s integration time, can be compared with pfd levels, also expressed in W/m<sup>2</sup> (defined assuming a 0 dBi receiving antenna gain in the direction of interference and given this integration time).

NOTE 1 – It is assumed that each transmitter is located in the far field of the radio telescope (that is, at a distance greater than  $2D^2/\lambda$ , where D is the effective diameter of the radio telescope and  $\lambda$  is the observing wavelength). Though this may not always be satisfied, it is considered to be an adequate approximation.

NOTE 2 – For some telescopes, the direction of maximum gain (boresight direction) may not always coincide with the geometrical axis of the radio telescope.

NOTE 3 – In the case of active antennas,  $P_i$  should be taken as the radiated RF power rather than the power at the input to the antenna.

NOTE 4 – The antenna gain of the transmitting station,  $G_t(\theta_i)$ , is taken at the frequency of the radio astronomy band considered. This may differ from the gain at the frequencies of the intended transmissions.

#### ANNEX 2

# **Model of radio telescope antenna pattern**

Antenna patterns, such as the one described in Recommendation ITU-R-SA.509, are not appropriate for use in a dynamic environment. In a dynamic environment, the model described in Recommendation ITU-R-S.1428 is used for FSS antennas. Further work is needed on the definition of radio astronomy antenna patterns. In the interim, and in the absence of measured patterns, the

Recommendation ITU-R S.1428 patterns may be considered as representative of radio astronomy antennas, for both the main beam and side lobe regions. The following example is extracted from Recommendation ITU-R S.1428 for the pattern for reflectors larger than  $100 \lambda$  in diameter:

$$G(\varphi) = G_{max} - 2.5 \times 10^{-3} \left(\frac{D}{\lambda} \varphi\right)^{2} \qquad \text{dBi} \qquad \text{for} \qquad 0^{\circ} \leq \varphi < \varphi_{m}$$

$$G(\varphi) = G_{1} \qquad \text{dBi} \qquad \qquad \text{for} \qquad \varphi_{m} \leq \varphi < \varphi_{r}$$

$$G(\varphi) = 29 - 25 \log \varphi \qquad \text{dBi} \qquad \qquad \text{for} \qquad \varphi_{r} \leq \varphi < 10^{\circ}$$

$$G(\varphi) = 34 - 30 \log \varphi \qquad \text{dBi} \qquad \qquad \text{for} \qquad 10^{\circ} \leq \varphi < 34.1^{\circ}$$

$$G(\varphi) = -12 \qquad \text{dBi} \qquad \qquad \text{for} \qquad 34.1^{\circ} \leq \varphi < 80^{\circ}$$

$$G(\varphi) = -7 \qquad \text{dBi} \qquad \qquad \text{for} \qquad 80^{\circ} \leq \varphi < 120^{\circ}$$

$$G(\varphi) = -12 \qquad \text{dBi} \qquad \qquad \text{for} \qquad 120^{\circ} \leq \varphi \leq 180^{\circ}$$

where:

$$G_{max} = 20 \log \left(\frac{D}{\lambda}\right) + 8.4 \qquad \text{dBi}$$

$$G_1 = -1 + 15 \log \left(\frac{D}{\lambda}\right) \qquad \text{dBi}$$

$$\phi_m = \left(\frac{20 \lambda}{D}\right) \sqrt{(G_{max} - G_1)} \qquad \text{degrees}$$

$$\phi_r = 15.85 \left(\frac{D}{\lambda}\right)^{-0.6} \qquad \text{degrees}$$

Alternately, a possibly more accurate representation for the innermost 1° of the pattern is given below, and may be used for this part of the antenna pattern.

#### 1 Model of main beam

A realistic approach is to use the following model for the main beam of a circular antenna:<sup>1</sup>

$$G_r(\varphi) = G_{r,max} \left[ \frac{J_1 (2\pi x)}{\pi x} \right]^2$$
 (3)

where:

$$G_{r,max} = \begin{bmatrix} 4\pi A_{eff} \\ \lambda^2 \end{bmatrix} : \text{maximum gain (expressed as a ratio)}$$

where:

 $A_{eff} = \pi (D/2)^2$ :area of the aperture of the telescope (m<sup>2</sup>)

D: effective diameter of the telescope (m)

 $x = \frac{\pi D \varphi}{360 \lambda}$ , with  $\varphi$  the off-boresight angle (degrees)

Equations (3) and (4) are expressed as a ratio and not in dB.

 $J_1(x)$ :1st order Bessel function.

The first null in this antenna pattern is at:

$$\varphi_0 = 69.88/(D/\lambda)$$
 degrees off-boresight

For example, if D = 100 m and  $\lambda = 3$  cm then  $G_{r,max} = 1.09 \times 10^8$  (equivalent to +80.4 dBi), and:

$$\varphi_0 = 0.0209 \qquad \text{degrees}$$

#### 2 Model of near side lobes up to 1° from the boresight

The following model is proposed for the near-in side lobes in the region  $\phi_0 \le \phi \le 1^{\circ}$ :2

$$G_r(\varphi) = B \left[ \frac{\cos(2\pi x - (3\pi/4) + 0.0953)}{\pi x} \right]^2$$
 (4)

where:

$$x = \frac{\pi D \varphi}{360 \lambda}$$
, with  $\varphi$  the off-boresight angle (degrees)

D: effective diameter of the radio telescope (m)

<del>λ: wavelength of operation (m)</del>

and

$$B = 10^{3.2} \pi^2 ((\pi D/2)/(180 \lambda))^2$$

# Annex <u>2</u>3

# Distribution of epfd levels

This Annex describes a way to derive epfd statistics over the whole sky.

### 1 Division of the sky into cells of approximately equal solid angle

The first step of this approach is to divide the sky into M rings parallel to the horizon and equally spaced in terms of elevation angle, from  $0^{\circ}$  to  $90^{\circ}$ . The width of each ring is  $90^{\circ}/M$ . The next step is to divide these rings into cells whose azimuth width is chosen to provide an integer number of cells per ring and is approximately equal to:

$$\frac{90/M}{\cos \text{ (elevation)}}$$
 degrees

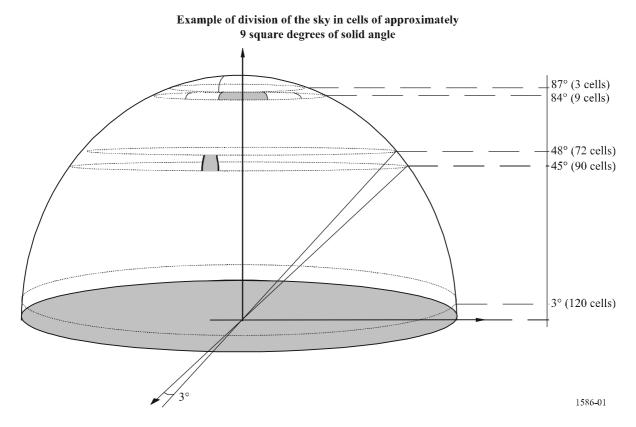
<sup>&</sup>lt;sup>2</sup> Equations (3) and (4) are expressed as a ratio and not in dB.

Figure 1 provides an example of division based on a step of 3° width in elevation, this divides the sky into 30 rings of 3° of elevation angle. Then, the azimuth width is approximately equal to:

$$\frac{90/30}{\cos \text{ (elevation)}}$$
 degrees

Elevation is a mean elevation in a given ring.

FIGURE 1



This leads to a division of the sky into 2 334 cells of approximately 9 square degrees of solid angle each. Table 1 provides the number of cells for each ring corresponding to this example.

TABLE 1

Example of division of the sky into square cells of about 9 square degrees solid angle

Lower elevation of the ring (degrees)	Ring solid angle (square degrees)	Cumulative solid angle (square degrees)	Azimuth step (degrees)	Number of cells in the ring	Cell solid angle (square degrees)	Cumulative number of cells	Percentage of solid angle (%)	Cumulative solid angle (%)
0	1 079.51	1 079.51	3	120	9	120	5.23	5.23
3	1 076.55	2 156.05	3	120	8.97	240	5.22	10.45
6	1 070.64	3 226.69	3	120	8.92	360	5.19	15.64
9	1 061.79	4 288.49	3	120	8.85	480	5.15	20.79
12	1 050.04	5 338.53	3	120	8.75	600	5.09	25.88
15	1 035.41	6373.93	3	120	8.63	720	5.02	30.90
18	1 017.94	7 391.87	3	120	8.48	840	4.94	35.84
21	997.68	8 389.55	3	120	8.31	960	4.84	40.67
24	974.68	9 364.23	3	120	8.12	1 080	4.73	45.40
27	949.01	10313.24	3	120	7.91	1 200	4.60	50
30	920.75	11 233.99	4	90	10.23	1 290	4.46	54.46
33	889.95	12 123.94	4	90	9.89	1 380	4.31	58.78
36	856.72	12980.66	4	90	9.52	1 470	4.15	62.93
39	821.14	13 801.81	4	90	9.12	1560	3.98	66.91
42	783.31	14 585.12	4	90	8.70	1 650	3.80	70.71
45	743.34	15 328.46	4	90	8.26	1 740	3.60	74.31
48	701.32	16029.79	5	72	9.74	1812	3.40	77.71
51	657.39	16 687.17	5	72	9.13	1 884	3.19	80.90
54	611.65	17 298.82	5	72	8.50	1956	2.97	83.87
57	564.23	17 863.06	6	60	9.40	2016	2.74	86.60
60	515.27	18 378.33	6	60	8.59	2076	2.50	89.10
63	464.90	18 843.23	6	60	7.75	2 136	2.25	91.35
66	413.25	19 256.48	8	45	9.18	2 181	2.00	93.36
69	360.47	19616.95	9	40	9.01	2 221	1.75	95.11
72	306.70	19 923.65	10	36	8.52	2 2 5 7	1.49	96.59
75	252.09	20 175.74	12	30	8.40	2 287	1.22	97.81
78	196.79	20 372.53	18	20	9.84	2307	0.95	98.77
81	140.95	20 513.49	24	15	9.40	2322	0.68	99.45
84	84.73	20 598.21	40	9	9.41	2331	0.41	99.86
87	28.27	20 626.48	120	3	9.42	2334	0.14	100

### 2 epfd distribution for a cell

First, a random choice is made for a pointing direction of the radio astronomy service antenna which will lie within a specific cell on the sky as defined in the paragraph above. Then, the starting time of the constellation is randomly chosen. The epfd is then evaluated for each time sample over a 2000 s integration time. The average epfd corresponding to this trial is then calculated for the chosen pointing direction and starting time of the constellation.

This operation is repeated to obtain a statistical distribution of the epfd in the considered cell. The methodology involves a number of trials, each of which calculates the averaged epfd level over a 2000 s integration interval. The greater the number of trials, the more accurate this distribution will be. A sufficient number of trials is needed to achieve the required confidence level in the results. In particular, the number of trials multiplied by the 2000 s integration time should be significantly higher than the period of the constellation. It is also necessary to ensure adequate statistical sampling over the full period of the constellation. Once it is found that no further significant change occurs in the distribution, it can be concluded that a sufficient number of trials has been performed. This check can be done either automatically as an integral part of the simulation, or manually, by stopping the simulation at regular intervals.

# **3** epfd distribution in worst-case pointing directions (to be applied only if the pfd levels from satellites are constant for a given elevation angle of radio astronomy service antenna)

The evaluation of the epfd distributions in cells on the sky may be simplified by first evaluating the epfd distribution in pointing directions corresponding to worst-case pointing directions. These worst-case pointing directions may be taken as those pointing directions where the probability of visibility of satellites is the highest. These pointing directions may be determined according to Recommendation ITU-R S.1257 – Analytical method to calculate short-term visibility and interference statistics for non-geostationary satellite orbit satellites as seen from a point on the Earth's surface (equations (28) and (29)). For a given elevation angle and a given constellation of non-GSO satellites, this Recommendation allows the calculation of the worst-case azimuths (there are usually two worst-case azimuths at a given elevation).

For the cells within which these worst-case pointing directions lie, the epfd distribution may be evaluated for a sufficient number of 2000 s integration times. Then, this epfd distribution may be compared with a pfd threshold level (defined assuming a 0 dBi receiving antenna gain in the direction of interference and given a 2000 s integration time).

For a cell, the percentage of time during which a pfd threshold level is exceeded can be calculated as the percentage of 2000 s integration periods in which the average pfd at the radio telescope exceeds this pfd threshold level.

Considering the 2% criterion in *recommends* **2** of Recommendation ITU-R RA.1513-1, the Comparison of the epfd distribution with the pfd threshold level for cells corresponding to these worst-case pointing directions, can provide the following conclusions:

- If, in all *M* rings, the time-averaged epfd for the worst-case pointing directions is equal to or less than the threshold of detrimental interference for 98% or more of time (see Fig. 2), then the criteria for avoidance of detrimental interference are met over the whole sky. If, in all the *M* rings, the pfd threshold is met in the worst-case pointing directions, during a percentage of time higher than the percentage of time criterion, this means that the *criteria* will be met over the whole sky.
- If, in a ring defined for a given elevation angle, the time-averaged epfd for the worst-case pointing directions is equal to or less than the detrimental threshold for 98% or more of time (see Fig. 2), then the criteria for avoidance of detrimental interference are met for the corresponding entire ring. If, in a ring defined for a given elevation angle, the pfd threshold is met in the worst-case pointing directions, during a percentage of time higher than the percentage of time criterion, this means that the criteria will always be met for the corresponding entire ring.
- If the interference criteria are not met, then further investigation is needed.

The consideration of these worst-case pointing directions provides information on the location of worst-case cells.

# 4 Output in terms of epfd distribution

The epfd calculation described in § 2 provides a distribution of epfd levels for each cell of the sky as shown in Fig. 2.

FIGURE 2 Example of a cumulative epfd distribution for a cell 100 90 Cumulative probability (%) 80 70 60 50 40 30 20 10 0 epfd 1586-02

FIGURE 2 Example of a cumulative epfd distribution for a cell

